



GNSS Clocks: Challenges and Developments

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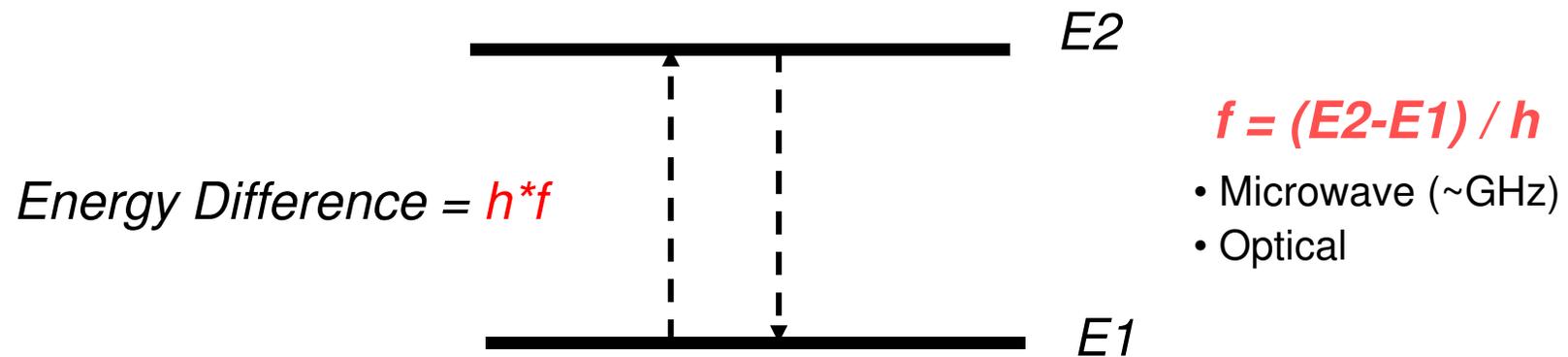
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Atomic Frequency Standards

Atomic clocks are based on an electron transitions, assumed *identical* everywhere.

*“Since 1967 the unit of the **second** is **defined** to be “the duration of 9,192,631,770 periods of the radiation corresponding to the transition between two hyperfine levels of the ground state of the cesium atom.”*



Ideal World: Nearly a perfect clock!

Real World: Environmental perturbations, measurement noise, & relativity.

Different atomic transitions have different benefits and implementation challenges.



GNSS Clock Technologies : Challenges and Developments

- Many atomic clock technologies, specialized to specific applications
 - Inherent atomic sensitivity
 - Localization method (cells, beams, RF traps, optical traps)
 - State selection method (magnetic selection, optical pumping)
 - Engineered isolation (if needed)
- Only a few meet the criteria for space operation: Rb (6.8 GHz), Cs(9.2 GHz), H(1.4 GHz)
 - Size, mass, and power constraints
 - Continuous, reliable, and long life operation
 - High immunity (or engineered isolation) to changing environments
- Current GNSS clocks
 - GPS: Rb, Cs
 - GLONASS Cs
 - GALILEO H, Rb
 - BEIDOU Rb
 - IRNSS Rb
 - QZSS Rb



Current GNSS Space Clock Technology

Industrialized Clocks (ground & space) : Rb, Cs, H

- Continuous, reliable, long life operation
- **Key Apps:** Telecommunications, timescale flywheels, space navigation.

Future GNSS Space Clock Technology?

Trapped Ion Frequency Standards (room temp)

- Practical operation with very good long term stability
- Amenable to lower SWaP
- **Key Apps:** Autonomous operation, Ultra stable timescales, space navigation

DSAC

Laser Cooled Standards – microwave (cold fountains, cold beams, cold ions)

- Laser cooled/trapped/interrogated atoms
- **Key Apps:** Accuracy metrology – definition of the second (Cs)

ACES, CAL

Laser Cooled Standards – optical (trapped ion or neutrals, optical lattices)

- Laser cooled/trapped/interrogated atoms
- Laser (optical) local oscillator. Count optical frequencies through femtosecond combs
- **Key Apps:** Accuracy metrology - future definition of the second

Chip Scale Atomic Clocks (CSAC): Rb, Cs

- MEMS fabrication and low power
- Large performance compromises
- **Key Apps:** Battery powered, requiring accuracy not achievable with quartz resonators.

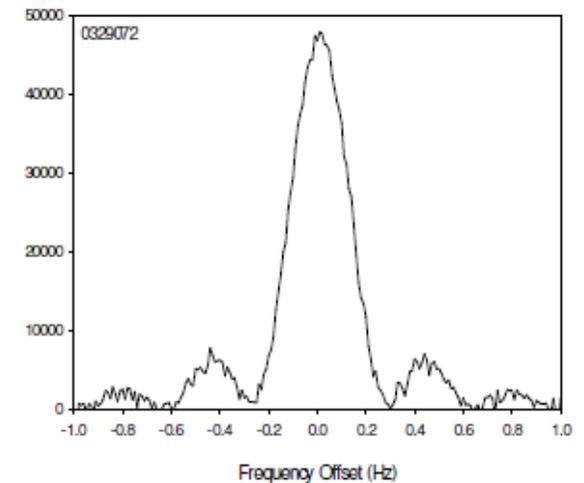
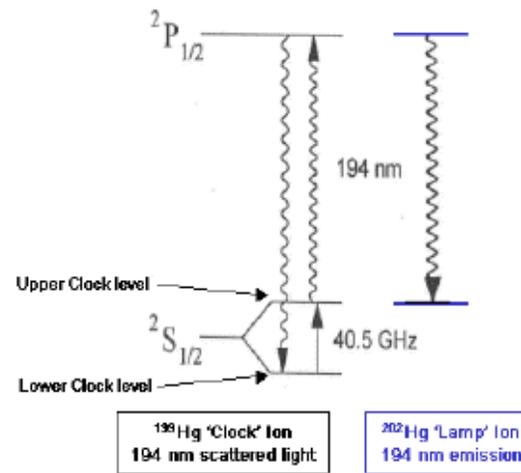
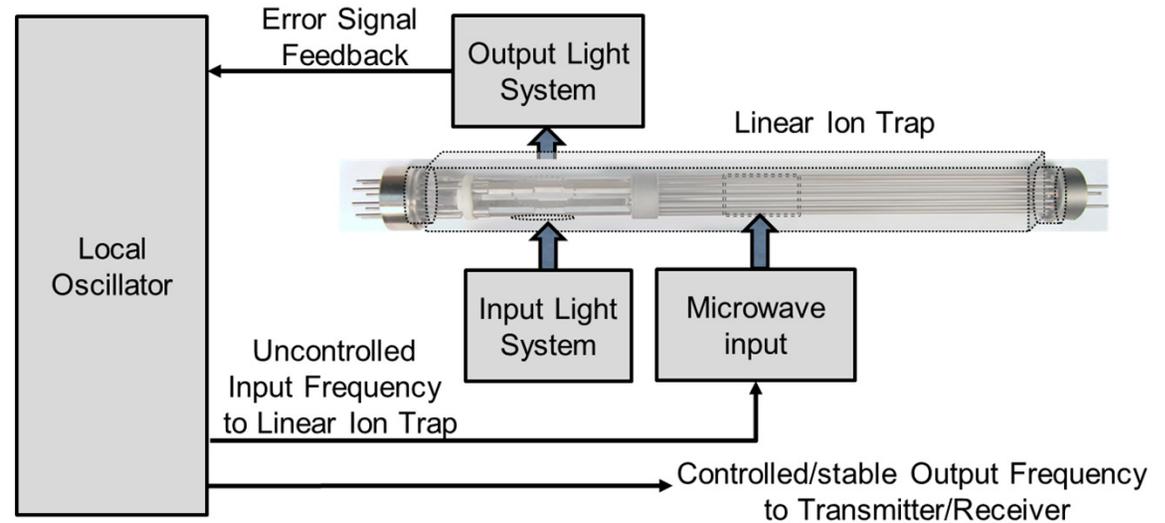


Key Performance Features:

- 10^6 - 10^7 $^{199}\text{Hg}^+$ trapped ions**
 - No wall collisions, high Q microwave line
 - buffer gas cooled to $\sim 300\text{K}$
 - multi-pole ion trap – **low T sensitivity**
- State selection:**
 - Optical Pumping from $^{202}\text{Hg}^+$ lamp
 - 1-2 UV photons per second scattered
- High Clock Transition:**
 - 40,507,347,996.8 Hz – **low B sensitivity**
- Adapts to variety of Local Oscillators - flexible**

Key Reliability Features: - practical

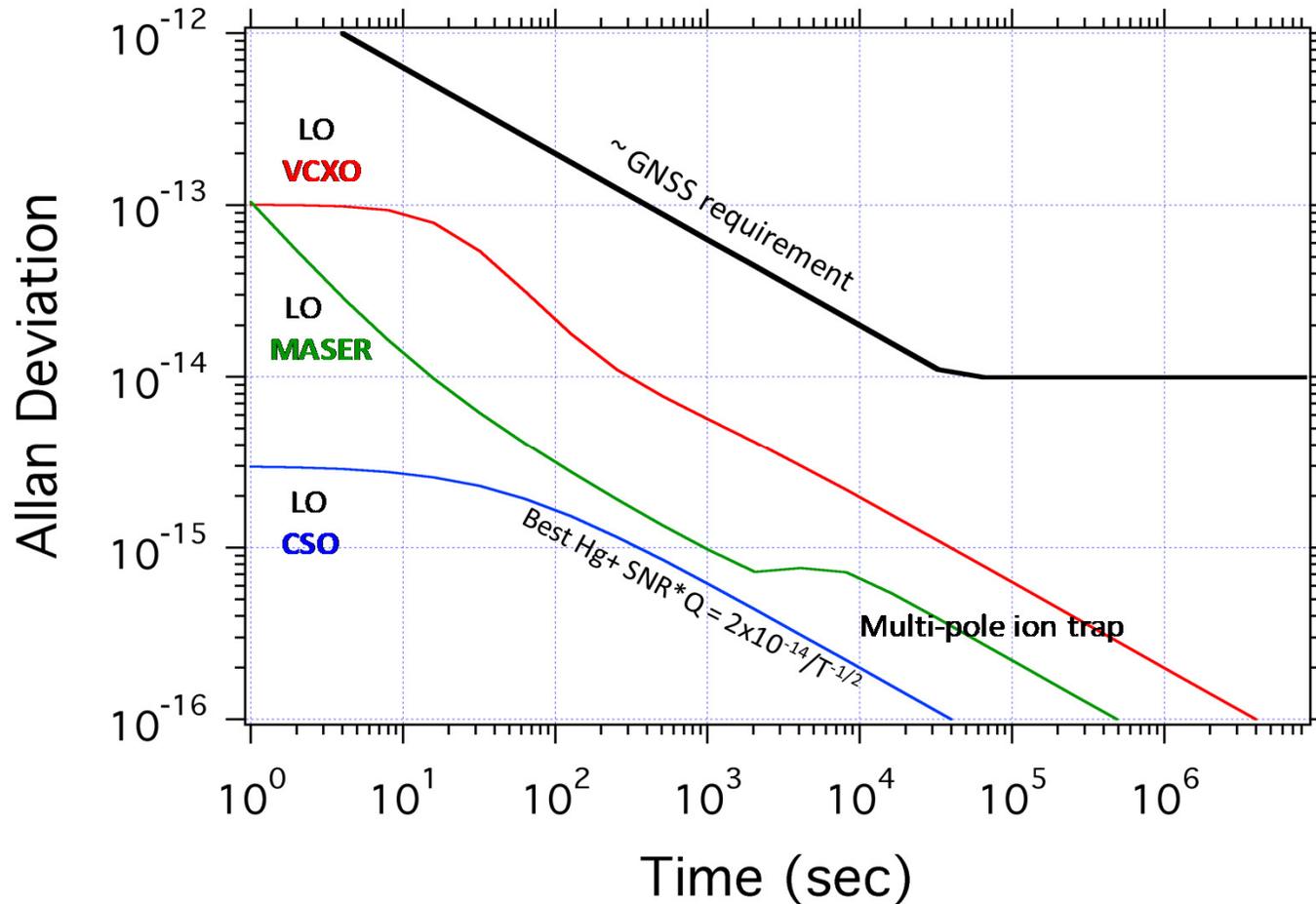
- No Lasers
- No Cryogenics
- No Microwave cavity
- No Light Shift
- Little/no Consumables





Hg+ Clock stability with differing Local Oscillators (LO)

- Short term stability depends on the LO
- Long term stability determined by the *Hg+* systematics



- Space clocks (e.g. GNSS) would be configured with a **USO LO**
- Optical-microwave LO being considered for advanced science missions.



- **Long life & continuous high stability operation**

Killer App: Ultra-stable timekeeping and long term autonomy.

Killer App: Amenable to small, low power operation- space.



Original LITS

- **Mercury Ion Clock Paths and Applications (2014):**

Ultra-Stable Performance: UTC timescales.

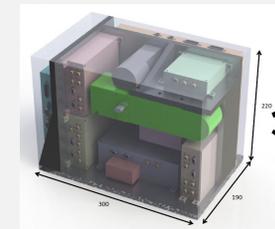
- “Compensated” Multi-pole ion trap.
- New References: drift $\leq 10^{-17}$ /day.
- 10^{-15} short term stability (~ 1 sec) via super LO’s.



Ultra stable ion clock – Chassis Size

Space Operation: Reliable, long life. USO LO.

- DSAC Technology Demonstration Mission (TRL 5-7)
- Deep Space: 3×10^{-13} short term, 10^{-14} at 1 day
- GNSS: 1×10^{-13} short term, 10^{-15} at 1 to 10 days



DSAC

Deep Space Infusion

GNSS Infusion

Ultra-small, low power ion clock technologies:

- Few cm^3 ion trap, Miniature light sources and LO’s under development.



$<10 \text{ cm}^3$



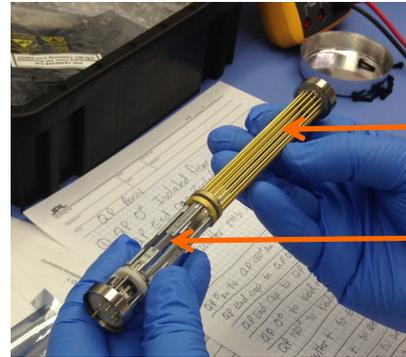
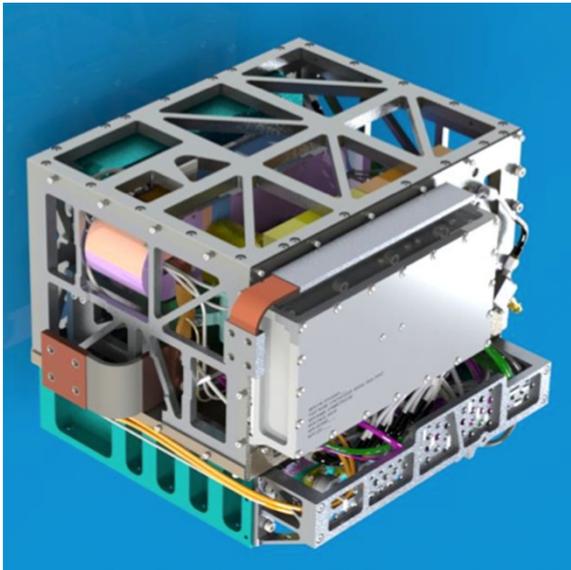
DSAC Technology Demonstration Mission (NASA TDM):

- First space demonstration of ion clock technology
- Establishes cross cutting path to Deep Space, GNSS, and Science Applications



NASA's DSAC Technology Demonstration Mission

DSAC Demonstration Unit (CAD)



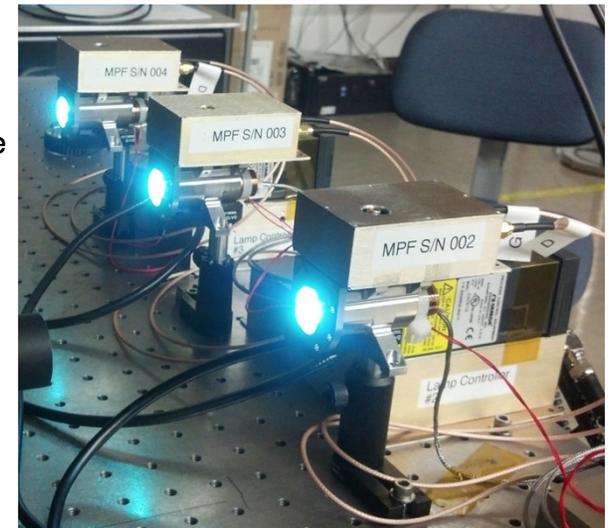
Vacuum Tube

Multi-pole Trap

Quadrupole Trap



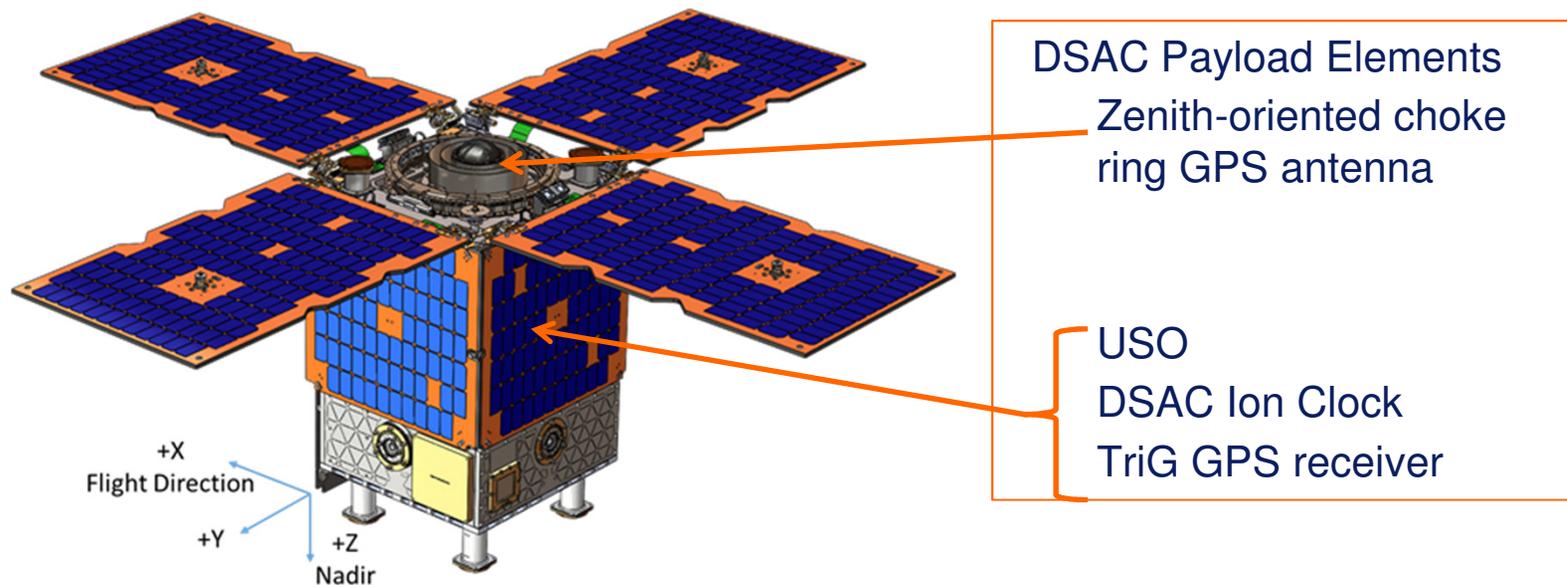
Mercury UV Lamp Testing



Develop “Demonstration Unit” mercury ion clock for navigation/science in deep space

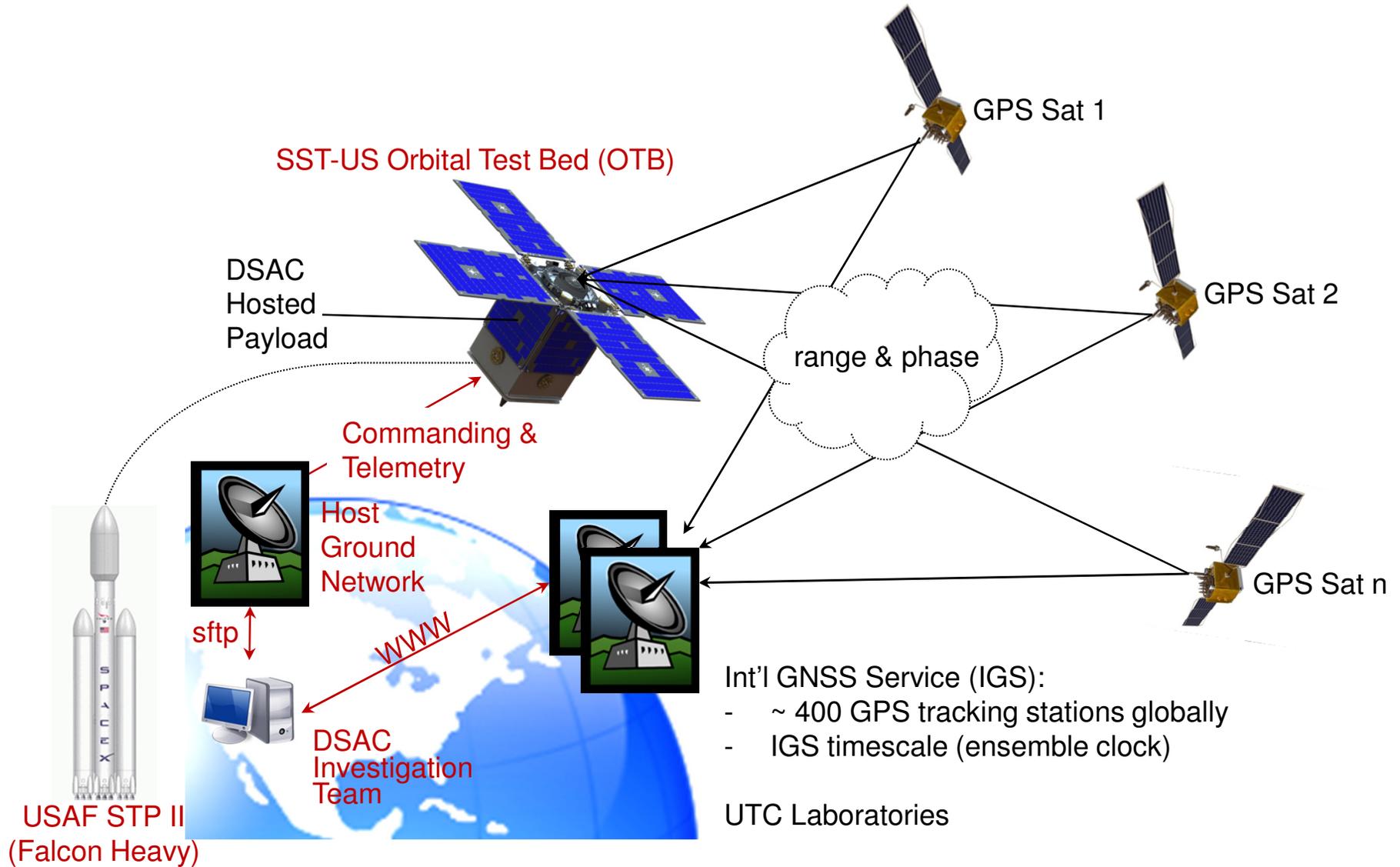
- Advance technology to TRL 7 and perform 1 year demonstration in space.
- Use commercial USO LO. Monitor long term stability through GPS Time Transfer
- Future operational unit (TRL 7 → 9) to be smaller, more power efficient.

DSAC Demonstration Payload and Hosting



- Flight experiment of the ion clock as a hosted payload on Surrey Satellite Technology US Orbital Test Bed II (OTB II) spacecraft
 - OTB II is a 180 kg ESPA-compatible spacecraft – fixed arrays, no active maneuvering, nadir fixed attitude maintained/controlled via reaction wheels/magnetorquers.
 - OTB II hosting other payloads including several US Air Force experiments
- Launched as part of USAF STP II (a Space X Falcon 9 Heavy). Scheduled for May 2016

DSAC Mission Architecture



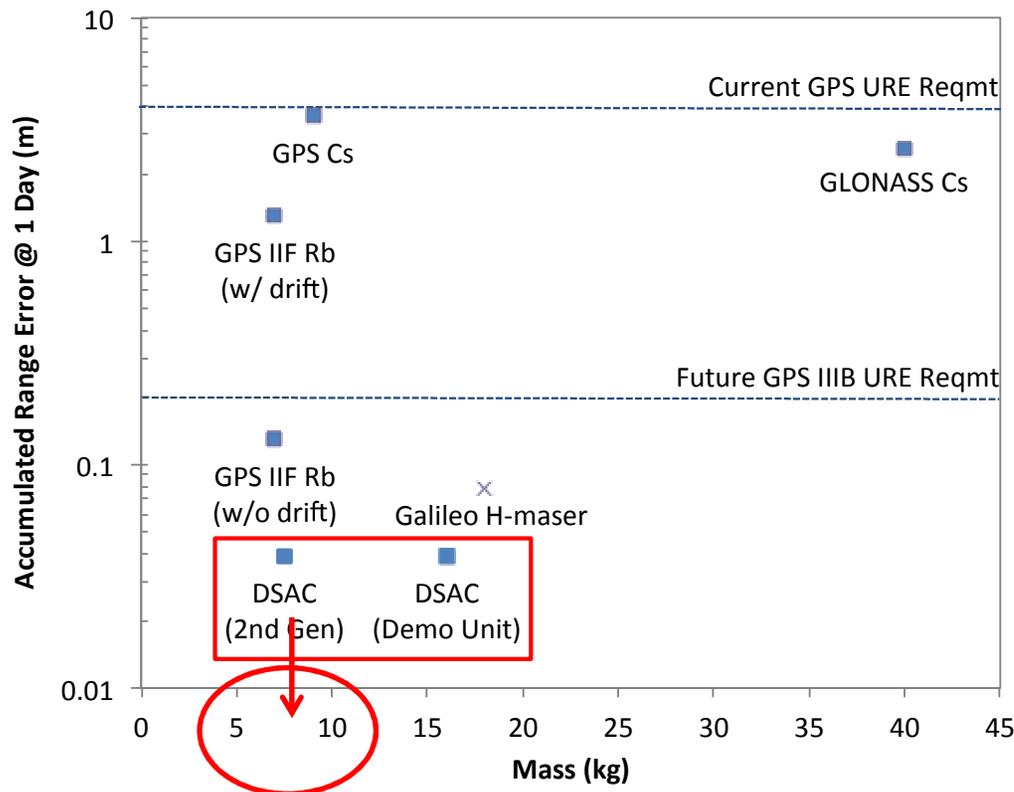


DSAC Mission Schedule

- Mission Definition & System Reqmts Review February 2012 ✓
- Preliminary Design Review May 2013 ✓
- NASA Commitment Review (KDP-C) November 2013 ✓
- **Clock Critical Design Review** July 2014
- Mission CDR & System Integration Review September 2014
- Pre-Ship Review March 2015
- Flight Readiness Review February 2016
- **Launch & Mission Operations** May 2016 + 1 Year

DSAC Compared to Existing GNSS Frequency Standards

- Required AD (including drift) of $< 3e-15$ at one-day (current estimate at $1.5e-15$)
- 2nd Generation DSAC will focus on packaging and lifetime.
- Satisfies future GPS IIIB URE requirement (includes clock and ephemeris)



AFS	Average Power
DSAC Demo Unit (1 st Generation)	< 50 W < 16 kg
DSAC Future Unit (2 nd Generation)	< 30 W
GPS IIF Rb (5 th Generation)	< 40 W
Galileo H-Maser (2 nd Generation)	< 60 W

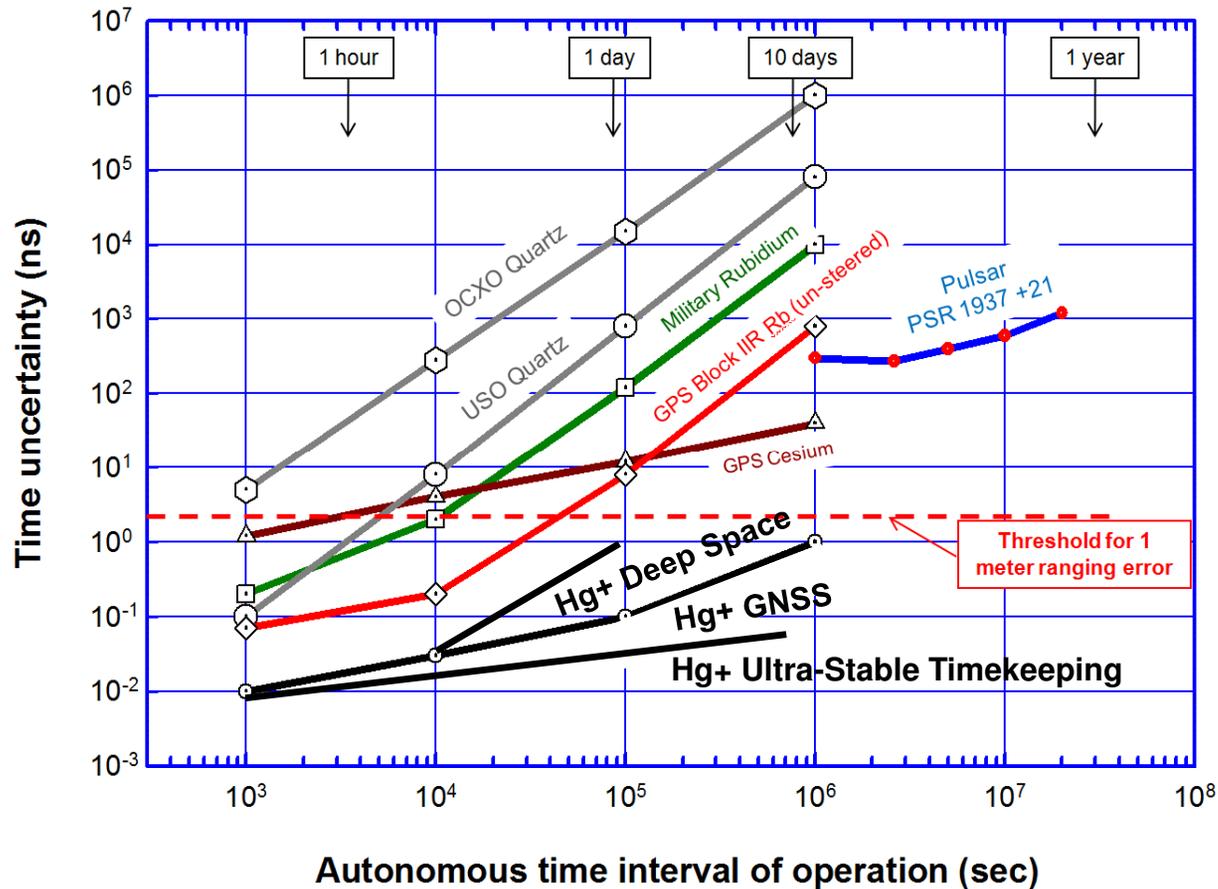
Further ion clock stability already demonstrated on ground if needed in space.

Mercury Ion Clock Timekeeping

Time Uncertainty after Unattended Operation versus Time

Ground Clock Stability < 0.1 ns at 10 days (10^{-16})

Space Clock Stability < 1 ns at 10 days (10^{-15})



Threshold for 1 meter ranging error

“A Compensated Linear Ion Trap Mercury Frequency Standard for Ultra-Stable Timekeeping”, *IEEE Trans of UFFC*, 55, No. 12, (2008).

“Mercury Atomic Frequency Standards for Space Based Navigation and Timekeeping”, *Proc. of PTTI* (2011)